

AD-A053 726

RICE UNIV HOUSTON TEX AERO-ASTRONAUTICS GROUP

F/G 20/11

SOME QUALITATIVE CONSIDERATIONS ON THE NUMERICAL DETERMINATION --ETC(U)

1977

A MANGIAVACCHI, A MIELE

AFOSR-76-3075

UNCLASSIFIED

AAM-WP-2

AFOSR-TR-78-0724

NL

[OF]

AD
A053726



END
DATE
FILMED

6 -78

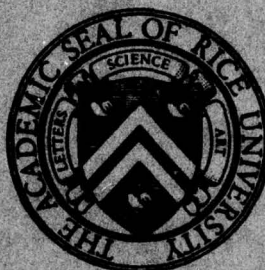
DDC

Ho
(2)

AD A 053726

AD No. _____
DDC FILE COPY

RICE UNIVERSITY



DISTRIBUTION STATEMENT A

Approved for public release;
Distribution Unlimited

AERO-ASTRONAUTICS MEMORANDUM NO. WP-2

Some Qualitative Considerations on
the Numerical Determination of Minimum Mass Structures
with Specified Natural Frequencies

by

A. Mangiavacchi and A. Miele

1977

DDC
REFORMED
MAY 10 1978
B

AERO-ASTRONAUTICS MEMORANDUM NO. WP-2

Some Qualitative Considerations on
the Numerical Determination of Minimum Mass Structures
with Specified Natural Frequencies

by

A. Mangiavacchi and A. Miele

RICE UNIVERSITY

1977

DISTRIBUTION STATEMENT A

Approved for public release;
Distribution Unlimited

DDC
RECEIVED
MAY 10 1978
B

Some Qualitative Considerations on
the Numerical Determination of Minimum Mass Structures
with Specified Natural Frequencies^{1,2}

by

A. Mangiavacchi³ and A. Miele⁴

Abstract. The problem of the axial vibration of a cantilever beam is investigated analytically. The range of values of the frequency parameter having technical interest is determined.

Key Words. Structural optimization, cantilever beams, axial vibrations, fundamental frequency constraint.

¹This research was supported by the Office of Scientific Research, Office of Aerospace Research, United States Air Force, Grant No. AF-AFOSR-76-3075.

²The authors are indebted to Dr. V.B. Venkayya, Wright-Patterson AFB, Ohio, for suggesting the topic.

³NATO Post-Doctoral Fellow, Department of Mechanical Engineering and Materials Science, Rice University, Houston, Texas.

⁴Professor of Astronautics and Mathematical Sciences, Rice University, Houston, Texas.

Section		<input checked="" type="checkbox"/>
Section		<input type="checkbox"/>
Section		<input type="checkbox"/>
DISTRIBUTION/AVAILABILITY CODES		
Dist.	AVAIL	and/or SPECIAL
A		

Notation

E	Modulus of elasticity, lb ft^{-2}
L	Length of the beam, ft
m	Normalized mass per unit length, $m = ML/M_0$
M	Mass per unit length, $\text{lb ft}^{-2} \text{ sec}^2$
M_0	Reference mass, $\text{lb ft}^{-1} \text{ sec}^2$ (Sections 2-3)
M_t	Tip mass, $\text{lb ft}^{-1} \text{ sec}^2$ (Sections 4-5)
M_*	Total mass of the beam, $\text{lb ft}^{-1} \text{ sec}^2$
x	Normalized axial coordinate, $x = X/L$
X	Axial coordinate, ft
u	Normalized axial displacement, $u = Y(X)/Y(L)$
Y	Axial displacement, ft
β	Frequency parameter, $\beta = \omega L / (\rho/E)$
ρ	Density, $\text{lb ft}^{-4} \text{ sec}^2$
ω	Natural frequency, sec^{-1}

Superscript

- ' Derivative with respect to the normalized axial coordinate x (for example, $u' = du/dx$)

1. Introduction

In this memorandum, we consider the problem of the axial vibration of a cantilever beam. With reference to a constant-section beam, we determine the range of values of the frequency parameter β having technical interest. This range of values of the frequency parameter is important in the solution of a subsequent problem: the determination of the mass distribution that minimizes the total mass of a beam for a given fundamental frequency constraint.

2. Nonoptimal Beam without a Concentrated Mass

Let m denote the normalized mass per unit length, u the normalized axial displacement, and β the frequency parameter. Let x denote the axial coordinate, normalized so that $x=0$ at the base of the beam and $x=1$ at the tip of the beam. Let the prime denote total derivative with respect to the axial coordinate x . With this understanding, the fundamental equation to be solved is the following:

$$(\mu')' + \beta^2 \mu = 0. \quad (1)$$

In this equation, the frequency parameter β is related to the natural frequency ω , the length L , the density ρ , and the modulus of elasticity E by the relation

$$\beta = \omega L \sqrt{\rho/E}. \quad (2)$$

In the absence of a concentrated mass attached at the tip of the beam, the boundary conditions for Eq. (1) are as follows:⁵

$$u(0) = 0, \quad m(1)u'(1) = 0. \quad (3)$$

If the mass distribution

$$m = m(x) \quad (4)$$

is prescribed a priori, then (1) is a second-order differential

⁵Equations (3) must be completed by the normalization condition $u(1) = 1$.

equation, to be solved in conjunction with the boundary conditions (3).

Constant Section. Next, we consider the particular case of a constant-section structure, that is, a structure with a constant mass per unit length:

$$m = \text{const.} \quad (5)$$

For this particular case, the differential equation (1) and the boundary conditions (3) simplify as follows:

$$u'' + \beta^2 u = 0, \quad (6)$$

$$u(0) = 0, \quad u'(1) = 0. \quad (7)$$

The solution of (6) consistent with the initial condition (7-1) is the following:⁶

$$u = A \sin(\beta x), \quad (8)$$

with the implication that

$$u' = A\beta \cos(\beta x). \quad (9)$$

From (9) and the final condition (7-2), we conclude that

$$\cos \beta = 0, \quad (10)$$

so that

$$\beta = (2n+1)\pi/2, \quad n = 0, 1, 2, \dots \quad (11)$$

⁶The constant A has the value $A = 1/\sin\beta$.

Therefore, for this problem, the smallest nontrivial value of the frequency parameter is

$$\beta = \pi/2 .$$

(12)

3. Optimal Beam without a Concentrated Mass

Now, suppose that a constant-section structure has been studied in accordance with Section 2. Suppose that the frequency parameter β which allows satisfaction of the boundary conditions (7) has been determined, namely, $\beta = \pi/2$. The total mass of the structure studied in Section 2 is given by⁷

$$M_*/M_0 = \int_0^1 m dx, \quad m = \text{const.} \quad (13)$$

Therefore, it is natural to pose the following question: for the same value of the frequency parameter $\beta = \pi/2$, is there a better beam, that is, one having a smaller total mass? In particular, is there a beam which yields the smallest total mass for the given value of β ? This question leads to the following variational problem: Minimize the total mass

$$M_*/M_0 = \int_0^1 m dx, \quad m = m(x), \quad (14)$$

with the understanding that the following constraints must be satisfied:⁸

$$(mu')' + \beta^2 mu = 0, \quad (15)$$

$$u(0) = 0, \quad m(1)u'(1) = 0, \quad (16)$$

and with the further understanding that $\beta = \pi/2$. Owing to

the fact that the problem (15)-(16) is homogenous, the obvious solution under the physical constraint

$$m(x) \geq 0 \quad (17)$$

is

$$m(x) = 0, \quad (18)$$

with the implication that

$$M_*/M_0 = 0. \quad (19)$$

In order to avoid the occurrence of the above trivial solution, Ineq. (17) could be changed as follows:

$$m(x) \geq m_0. \quad (20)$$

Then, the solution would become

$$m = m_0. \quad (21)$$

To arrive at solutions other than constant mass solutions, it is necessary to postulate some different physical situation (e.g., a concentrated mass attached at the end of the beam). In turn, this results in a change in the boundary condition (16-2), and this change makes it unnecessary to employ inequality constraints of the form (17) or (20).

⁷The symbol M_0 denotes a reference mass.

⁸Equations (16) must be completed by the normalization condition $u(1) = 1$.

4. Nonoptimal Beam with a Concentrated Mass

In this section, we assume that a concentrated mass M_0 is attached at the tip of the beam. Using the same terminology as in Section 2, we see that the governing differential equation (1) still holds:

$$(\mu')' + \beta^2 \mu = 0. \quad (22)$$

On the other hand, the boundary conditions (3) are modified as follows:⁹

$$u(0) = 0, \quad m(1)u'(1) = \beta^2. \quad (23)$$

Constant Section. Again, we consider the particular case of a constant-section structure. Under condition (5) and after observing that

$$M_*/M_0 = m, \quad (24)$$

then problem (22)-(23) becomes

$$u'' + \beta^2 u = 0, \quad (25)$$

$$u(0) = 0, \quad u'(1) = (M_0/M_*)\beta^2. \quad (26)$$

The solution of (25) consistent with the initial condition (26-1) is the following:

⁹Equations (23) must be completed by the normalization condition $u(1) = 1$.

$$u = A \sin(\beta x), \quad (27)$$

with the implication that

$$u' = A\beta \cos(\beta x). \quad (28)$$

From (28) and the final condition (26-2), we conclude that

$$A \cos \beta = (M_O/M_*)\beta. \quad (29)$$

Owing to the fact that

$$u(1) = A \sin \beta, \quad (30)$$

elimination of A from (29)-(30) leads to the following transcendental equation:

$$\beta \tan \beta = (M_*/M_O)u(1), \quad (31)$$

which, for $u(1)=1$, reduces to

$$\beta \tan \beta = M_*/M_O. \quad (32)$$

This equation supplies the frequency parameter β in terms of the mass ratio (ratio of beam mass M_* to tip mass M_O).

In order to understand the significance of (32), let us consider two limiting cases: (i) negligible mass ratio and (ii) infinite mass ratio. If $M_*/M_O = 0$, then the solution of (32) is

$$\beta = n\pi, \quad n = 0, 1, 2, \dots \quad (33)$$

On the other hand, if $M_*/M_0 = \infty$, then the solution of (32) is

$$\beta = (2n+1)\pi/2, \quad n = 0, 1, 2, \dots, \quad (34)$$

which is identical with (11). Since the first natural frequency corresponds to $n = 0$, we conclude that, for mass ratios in the range

$$0 \leq M_*/M_0 \leq \infty, \quad (35)$$

the smallest frequency parameter β consistent with the transcendental equation (32) lies in the range

$$0 \leq \beta \leq \pi/2. \quad (36)$$

5. Optimal Beam with a Concentrated Mass

As in Section 3, we can formulate the problem of finding the optimal mass distribution. The problem is as follows:
Minimize the total mass

$$M_*/M_0 = \int_0^1 m dx, \quad m = m(x), \quad (37)$$

with the understanding that the following constraints must be satisfied:¹⁰

$$(mu')' + \beta^2 mu = 0, \quad (38)$$

$$u(0) = 0, \quad m(1)u'(1) = \beta^2, \quad (39)$$

and with the further understanding that the frequency parameter β has some fixed value in the range

$$0 \leq \beta \leq \pi/2. \quad (40)$$

¹⁰

Equations (39) must be completed by the normalization condition $u(1) = 1$.

References

1. TURNER, M.J., Design of Minimum Mass Structures with Specified Natural Frequencies, AIAA Journal, Vol. 5, No. 3, 1967.
2. LEITMANN, G., An Introduction to Optimal Control, McGraw-Hill Book Company, New York, New York, 1966.
3. MIELE, A., Editor, Theory of Optimum Aerodynamic Shapes, Academic Press, New York, New York, 1965.
4. MIELE, A., PRITCHARD, R.E., and DAMOULAKIS, J.N., Sequential Gradient-Restoration Algorithm for Optimal Control Problems, Journal of Optimization Theory and Applications, Vol. 5, No. 4, 1970.
5. MIELE, A., IYER, R.R., and WELL, K.H., Modified Quasi-linearization Algorithm and Optimal Initial Choice of the Multipliers, Part 2, Optimal Control Problems, Journal of Optimization Theory and Applications, Vol. 6, No. 5, 1970.

UNCLASSIFIED

SECRET CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE

READ INSTRUCTIONS
BEFORE COMPLETING FORM

1. REPORT NUMBER

AFOSR-TR-78-0724

2. GOVT ACCESSION NO.

3. REPORTING OR CONTRACT NUMBER

AAM-WP-2

4. TITLE (and Subtitle)

SOME QUALITATIVE CONSIDERATIONS ON THE
NUMERICAL DETERMINATION OF MINIMUM MASS
STRUCTURES WITH SPECIFIED NATURAL
FREQUENCIES

5. TYPE OF REPORT & PERIOD COVERED

Interim Repts

6. PERFORMING ORG. REPORT NUMBER

Aero-Astro Memo WP-2

7. CONTRACT OR GRANT NUMBER(s)

AFOSR-76-3075

9. PERFORMING ORGANIZATION NAME AND ADDRESS

Rice University
Department of Mechanical Engineering
Houston, Texas 7700110. PROGRAM ELEMENT, PROJECT, TASK
AREA & WORK UNIT NUMBERS61102E
2304A3

11. CONTROLLING OFFICE NAME AND ADDRESS

Air Force Office of Scientific Research / M
Bolling AFB, DC 20332

12. REPORT DATE

1977

13. NUMBER OF PAGES

14 17 p

14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)

15. SECURITY CLASS. (of this report)

UNCLASSIFIED

15a. DECLASSIFICATION/DOWNGRADING
SCHEDULE

16. DISTRIBUTION STATEMENT (of this Report)

Approved for public release, distribution unlimited.

17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)

18. SUPPLEMENTARY NOTES

19. KEY WORDS (Continue on reverse side if necessary and identify by block number)

Structural optimization, cantilever beams, axial vibrations,
fundamental frequency constraint.

20. ABSTRACT (Continue on reverse side if necessary and identify by block number)

The problem of the axial vibration of a cantilever beam is inves-
tigated analytically. The range of values of the frequency para-
meter having technical interest is determined.

402 169

alt